# EXPERIMENTAL AND FINITE ELEMENT ANALYSIS OF UNIDIRECTIONAL AND BIDIRECTIONAL FIBRE REINFORCED COMPOSITE CANTILEVER PLATES

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## ABSTRACT

In vibration analysis mechanical resonance is a significant term. Resonance occurs when the frequency of its oscillations matches with the system's natural frequency of vibration. It may cause extreme swaying motion and the Structural failure in offensively constructed structures like bridges, buildings and airplane. This type of failure is known as resonance disaster. To avoid structural damage caused due to resonance, it is important to determine: 1. The natural frequencies to avoid resonance 2. The mode shapes.

The generally steel components are converse in nature and difficult for loading due to heavy well and transportation of heavy structures are also difficult so there is need to reprize the materials by new materials.

INDEX TERMS: Natural from your Damping, etc.

## INTRODUCTION

Composite n d as mixing of al are wellials (reinfor two or more than two lers, binder etc.) different in composition mall quant nposite e made from or more materia than two erent chemical and constituen materials with s, that when mix l make a material physical prodifferent from the ividual cor hts. There are many cases; the reinforce is tougher, stronger, harder and stiffer than the matrix. sein orcement is will be a fiber or a particulate. Particula composites have almost same dimensions in all directions. Particulate composites are less stiff as compared continuous fiber composites and also they are less expensive. Particulate reinforced composites usually contain less reinforcement upto 40 to 50 percent. A composite has technologically advanced speedily in the earlier 30 years through the improvement of fibrous composites that is in glass fiber reinforced polymers (GFRP) and, more recently, carbon fiber reinforced polymers (CFRP). Their growing use in ground transport systems is an replacement of metals to increase

usage which is set accelering. It finds application in composite, Autom tive, sport grow, medical equipment & packaging Lebustry.

stinness of the compos n be changed by sequence and fiber orientation. ch allows for stack Itering of the aterial to achieve the desired natural tł ies and tive mode shapes without changing increasing ts weight. Their intrinsic its ge lows the design to alter the material in anisotrop the desir a performance requirements. er to ac s, it is an in nt t develop a tool that will allow the otimized designs considering the gner to obta uctural requirements and functional characteristics. This work considers the behaviour of components anufactured from fiber reinforced composite materials. ccomplish this, some plates are taken which are made moulding process. Experimental tests are carried asing specimens with different thicknesses. From the out sults, the effect of the fibers orientations & number of layers on the natural frequencies and modal damping are investigated. These experiment results are used to validate the FEA results obtained from the ANSYS. The FEA and Experimental results are compared to analyze the composite plates.

Now a day's long fiber polymer are widely used in industries like aircraft and wind turbine components. Long fiber components are manufactured from carbon or glass fiber embedded in polymer like epoxy resin. Composite materials are ideal for structural application where ratio of high strength to weight and stiffness to weight are required. Aerospace applications are typical weight sensitive structures in which composite materials are cost effective. The study of composite materials involves many topics for example manufacturing processes, anisotropy, and elasticity strength of anisotropic materials. The main material properties for usual engineering mechanics applications are strength and stiffness. The fibers have high stiffness and have high strength and they carry the load to which the structure is submitted. The objective of this dissertation is to analyzer experimentally and by finite

element analysis the mechanical behaviour of composite material used for cantilever applications.

## FREQUENCIES OBTAINED BY FEA

Table: showing comparison of FEA results of natural frequency of glass fiber cantilever composite plates of various thicknesses and different orientation.

Below table shows that the comparison of natural frequencies obtained by the FEA of composite cantilever plates of Unidirectional fiber orientation and Bidirectional fiber orientation and the values of natural frequency are increasing as the plate thickness goes on increasing.

## **EXPERIMENTAL RESULTS**

### Experimentation is conducted to determine

- i) Natural Frequency and
- ii) Mode shapes of composite cantilever pates



### Fig: Experimental S

- Equipment used to determine a tural Frequency are
  - 1. Signal Analyser (F
  - 2. Accelerometer
  - 3. Impact Ham
  - Equipment for fip any de shapes by the rimentation 1. Signal Power Osciertor
    - Signal Power Os
      Exerter

Table: a wing comparison to previmental results of natural frequency of glass fiber call over composite plates of various the masses and different operation.

Modes	Unidirectiona			directional fiber		
.No	orientation(150			orientation(150*80)		
	10mm	5mm		10mm	5mm	2mm
1	384.23	192.13	4.03	704.65	384.38	128.1
		•				
5	2944.3	1792.6	704.4	3712.2	2432.1	1920.1
9	6144.2	3904.6	1856.1	9984.3	6656.5	4416.7

From above table natural frequency values for Bidirectional composite material are more as compared to unidirectional composite material. There is a difference in the values of natural frequency. Fundamental natural frequency of bidirectional composite is better than unidirectional composites.

#### **RESULTS AND COMPARISONS**

As Bidirectional fiber orientation composite cantilever plates shows higher range of natural frequencies we select Bidirectional fiber oriented composite plate as alternative material for steel components.

Table: shows that the imparison of Bidirectional fiber orientation 80x150x1 mm plate by FFT and FEA method.

methou								
Mode	Natural		%Difference	Average				
No.	Frequency	Freque	Difference					
	by FEA	by F.						
	Hz	analyzer in						
		Hz						
1	1.8	704.65	7.5					
5	6671.3	3712.2	0.3994	2.76%				
9	10025.3	9984.3	0.4					

plate by the comparison of 80x150x10mm plate by the and FEA method. The average difference of 76% is on used for the same plate.

Mode uses are some closer with the respective uencies by examined method as well as FEA as shown ig below.



Fig: Mode shape using Exciter



Fig: Mode shape using FEA

Experimental results of natural frequencies determined experimentally are vary than those obtained by FEM analysis due to certain parameters are as follows 1. Weight of accelerometer mounted on plate. 2. Damping caused at fixed end.

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3. Rotation of plates during flexural deformation.

- 4. Double hit of the hammer.
- 5. Manufacturing technique
- 6. Variation in the thickness
- 7. Non uniform surface finish
- 8. Bubbles

## CONCLUSION

1. Natural frequency increases as thickness of plate increases.

2. Natural Frequency shows better result for Bi directional fiber orientation.

3. Experimental natural frequency of glass fiber rectangular cantilever composite plates matches with the ANSYS results with avg. 2.76% difference.

4. Experimental results are matching with the FEM results hence we can replace Bidirectional glass fiber composite cantilever plate in place of steel plates.

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